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2005 JAWAHARLAL NEHRU TECHNOLOGICAL UNIVERSITY**IV B.TECH II SEMESTER SUPPLEMENTARY EXAMINATIONS
FATIGUE AND FRACTURE MECHANICS
(AERONAUTICAL ENGINEERING)**

JULY- 2005

TIME: 3 HOURS

MAX MARKS: 80

**Answer any FIVE Questions
All Questions carry equal marks**

1. (a) How the properties of materials and surface treatments improve the fatigue life of components?
2. (b) Explain how shot peening improves fatigue life.
2. (a) The fatigue limit of a 1045 steel is about 300 MN/m² when the mean stress is zero. The tensile strength of this steel is 750 MN/m². Estimate the safe stress amplitude for this material when the mean stress is 250 MN/m².
- (b) Discuss a typical fatigue testing procedure. Explain how the effect of mean stress can be studied.
3. (a) Explain the reasons for well defined fatigue limit in certain materials.
- (b) The endurance limit of a steel member is 112 Mpa and the tensile strength is 385 MPa. What is the fatigue strength corresponding to a life of 70×10^3 cycles.
4. (a) Describe about stress fluctuations and cumulative damage in fatigue failure.
- (b) How cumulative fatigue is expressed?
- (c) Discuss woods theory of fatigue failure.
5. (a) Explain the mechanism of fatigue failure propagation.
- (b) Discuss the interactions between parallel edge dislocations.
6. (a) A relatively large plate of glass is subjected to a tensile stress of 40MPa. If the specific surface energy and modulus of elasticity for this glass are 0.3 j/m² and 69 Gpa, respectively, determine the maximum length of a surface flaw that is possible without fracture.
- (b) What advantage does the fracture mechanics approach afford the engineer in designing components compared to more traditional approaches.
7. (a) Explain the differences in terms of work of brittle fracture between glass and a metal? Explain.
- (b) A metallic plate of dimensions 2mm×50mm×200mm, with one internal flaw (size = 1mm) is loaded in tension. If its fracture load is 2500 kg, calculate work of fracture (G_c); and critical stress intensity factor (K_{Ic}), Assume plane stress condition. The modulus of the material is 30 GPa. 8. Identify and explain several problems a designer must recognize when dealing with fatigue loading as compared with static loading.